

Encapsulated Solid Propellant Pulse Rockets

FOR

Extra
Vehicular
Activity



CURTISS-WRIGHT CORPORATION ■ WRIGHT AERONAUTICAL DIVISION
WOOD-RIDGE, NEW JERSEY U.S.A.

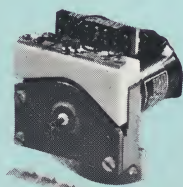
Encapsulated Solid Propellant



The ESPPR concept has been proven during five years of development. Several unique features are particularly applicable to EVA devices.

- SAFE ON-BOARD STORAGE
- INHERENT RE-LOAD CAPABILITY
- ROCKET MOTOR REDUNDANCY
- COMPLETE ON-BOARD SYSTEM VERIFICATION
- PRECISE PROPELLANT AVAILABILITY READ-OUT

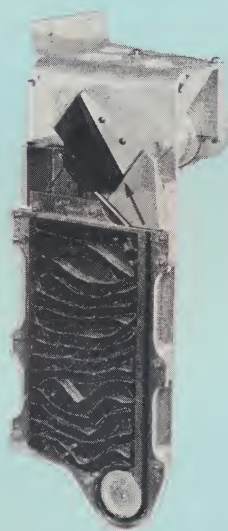
ESPPR Hardware



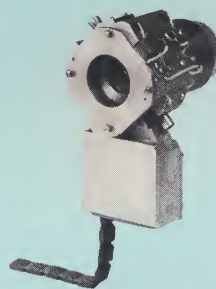
WSR-1
FEASIBILITY
MODEL



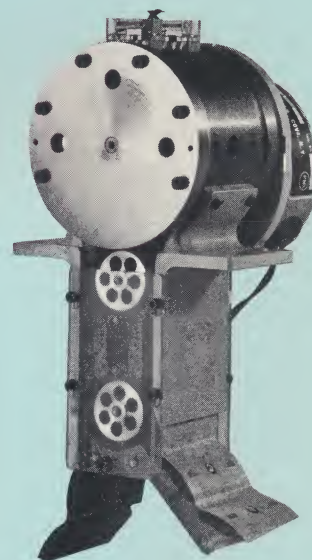
WSR-2
PROTOTYPE



WSR-2
FLIGHT SYSTEM



WSR-4
FOUR POSITION
E-DOME PROTOTYPE



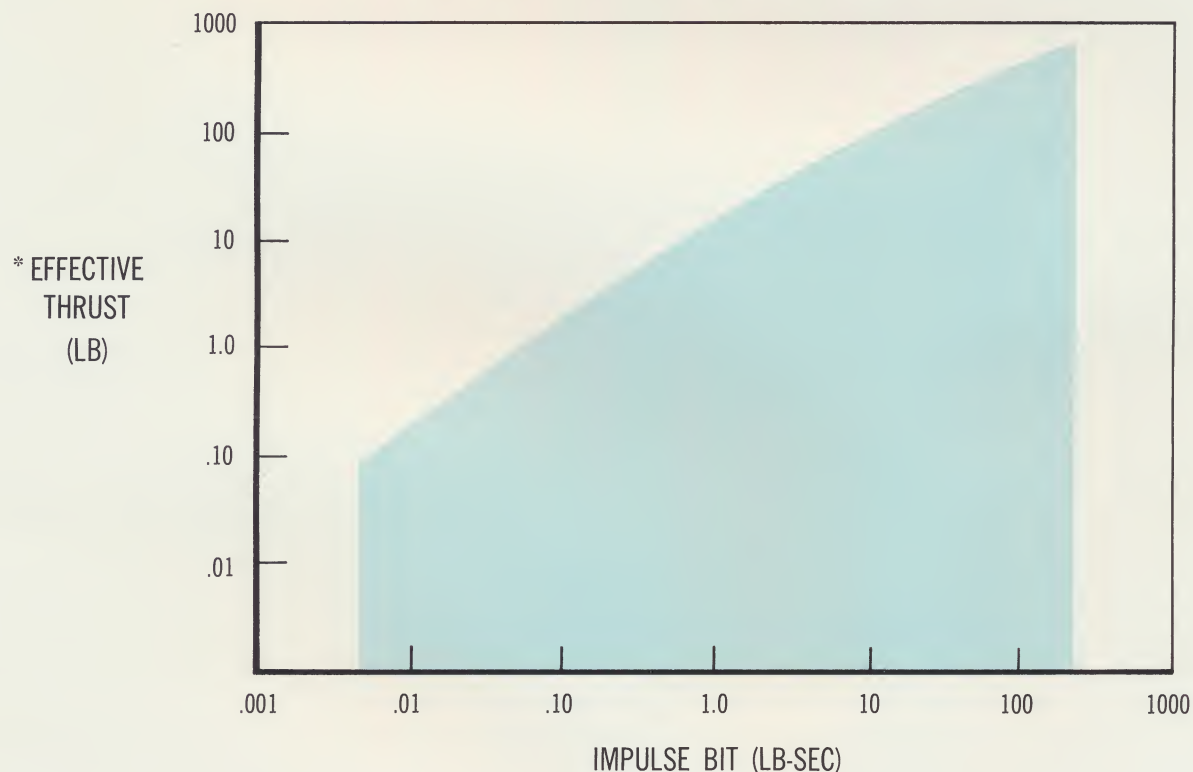
WSR-101
CAP CANNON PROTOTYPE

ant Pulse Rockets (ESPPR)

Test Experience

TOTAL FIRINGS	13,500
REPETITIVE FIRINGS	2,500
VACUUM	10^{-8} TORR FOR 30 DAYS
TEMPERATURE	-30°F TO $+180^{\circ}\text{F}$
VIBRATION	10 TO 2000 CPS. 3 TO 27g
RANDOM NOISE	20 TO 2000 CPS. 15 TO 22g
ACCELERATION	30g
IMPULSE BIT REPEATABILITY	$\sigma = 1.85\%$ FOR 100 FIRINGS
HIGH TEMPERATURE	NO FIRE AT 600°F
RADIATION	3.65×10^6 RADS
RANGE SAFETY	1 AMP. 5 MIN., NO FIRE
NON-JAMMING	10 CONSECUTIVE "BLOWS"
STORAGE SAFETY	FIRE CAPSULE IN LOADED MAGAZINE

ESPPR - Size Range



* Effective Thrust = Impulse per cap \times number of caps fired per second.

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